

Subject: Introduction to Solid Mechanics

Topic: Assignment - II Solution

Instructor: Prof. RASHID MUSTAFA

Q-1 (c)

$$\text{Hoop stress } (\sigma_h) = \frac{p d}{2t}$$

$$100 = \frac{2 \times 800}{2 \times t}$$

$$t = 8 \text{ mm}$$

Q-2 (b)

$$\epsilon_L \text{ (longitudinal strain)} = \frac{\sigma_L}{E} - \nu \frac{\sigma_h}{E}$$

$$= \frac{p \cdot d}{4tE} - \nu \frac{p d}{2tE}$$

$$= \frac{p d}{tE} \left[\frac{1}{4} - \frac{\nu}{2} \right]$$

$$\epsilon_L = \frac{p d}{4tE} (1 - 2\nu)$$

$$\text{Hoop strain } (\epsilon_h) = \frac{\sigma_h}{E} - \nu \frac{\sigma_L}{E}$$

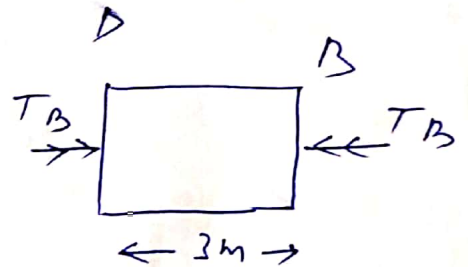
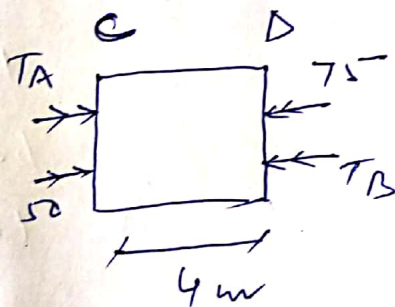
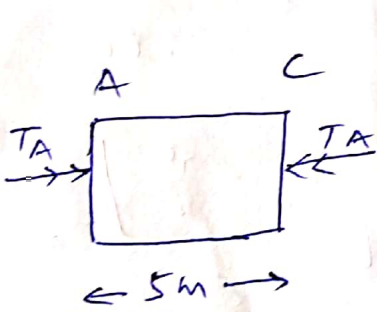
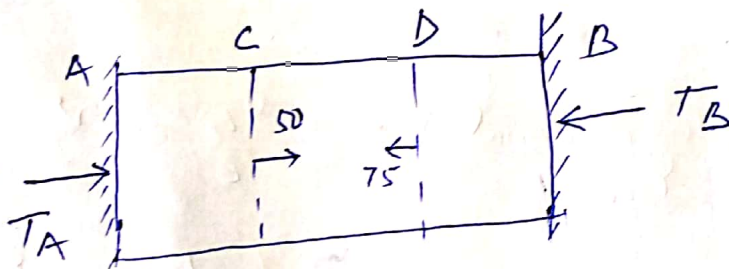
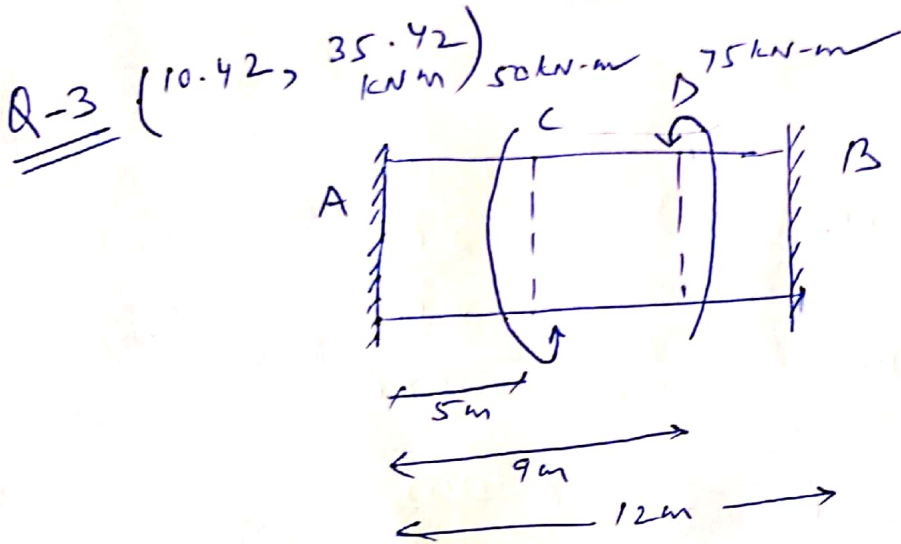
$$= \frac{p \cdot d}{2tE} - \nu \frac{p d}{4tE}$$

$$= \frac{p d}{t \cdot E} \left[\frac{1}{2} - \frac{\nu}{4} \right]$$

$$\frac{p.d}{4tE} [2-\nu]$$

(2)

$$\frac{E_L}{E_L} = \frac{\frac{p.d}{4tE} (1-2\nu)}{\frac{p.d}{4tE} (2-\nu)} = \frac{1-2\nu}{2-\nu}$$



Consider section CD

$$50 + T_A = T_B + 75$$

$$T_A - T_B = 25 \quad \text{--- (1)}$$

Since end A and B are fixed so,

$$\theta_{AB} = 0, \quad \theta_{AB} = \text{Angle of twist b/w A and B.}$$

$$\theta_{AB} = \theta_{AC} + \theta_{CD} + \theta_{DB}$$

$$\theta_{AB} = \theta_{AC} + \theta_{CD} + \theta_{DB}$$

$$\theta_B - \theta_A = \theta_{AC} + \theta_{CD} + \theta_{DB}$$

$$0 = \theta_{AC} + \theta_{CD} + \theta_{DB}$$

Using $\theta = \frac{TL}{NIP}$

$$\frac{T_A \times 5}{NIP} + \frac{4(50 + T_A)}{NIP} + \frac{3T_B}{NIP} = 0$$

$$5T_A + 200 + 4T_A + 3T_B = 0$$

$$9T_A + 3T_B = -200 \quad \text{--- (2)}$$

Solving Equation (1) & (2)

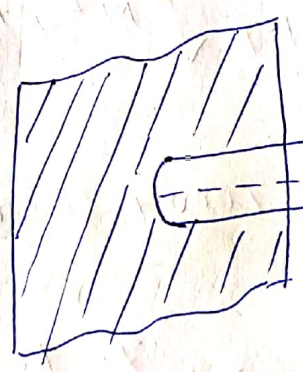
$$T_B = -35.42 \text{ kN-m}$$

$$T_A = -10.42 \text{ kN-m}$$

$$T_A = 10.42 \text{ kN-m}$$

$$T_B = 35.42 \text{ kN-m}$$

Q-4 (1648.95, -629.84, 1139.4)



Bending moment (M) = 1000 kg-m

Torque (T) = 2000 kg-m

Diameter of shaft = 10 cm = 0.1 m

Alc to Torsion Equation $\left[\frac{T}{IP} = \frac{\tau}{r} = \frac{\tau_{max}}{R} = \frac{N\theta}{L} \right]$

$$\tau_{max} = \frac{16T}{\pi D^3} = \frac{16T}{\pi \Delta^3}$$

A/c to Bending Equation

$$\frac{M}{I} = \frac{\sigma}{y} = \frac{E}{R}$$

$$\sigma = \frac{My}{I} = \frac{32M}{\pi D^3}$$

Let σ_{P1}/σ_{P2} be the Major and Minor Principal stress.

$$\sigma_{P1}/\sigma_{P2} = \frac{\sigma + 0}{2} \pm \sqrt{\left(\frac{\sigma - 0}{2}\right)^2 + \tau^2}$$

$$= \frac{\sigma}{2} \pm \sqrt{\left(\frac{\sigma}{2}\right)^2 + \tau^2}$$

$$\sigma_{P1}/\sigma_{P2} = \frac{32M}{2 \cdot \pi D^3} \pm \sqrt{\left(\frac{32M}{2 \cdot \pi D^3}\right)^2 + \left(\frac{16T}{\pi D^3}\right)^2}$$

$$= \frac{16M}{\pi D^3} \pm \sqrt{\left(\frac{16M}{\pi D^3}\right)^2 + \left(\frac{16T}{\pi D^3}\right)^2}$$

$$= \frac{16}{\pi D^3} \left[M \pm \sqrt{M^2 + T^2} \right]$$

Major Principal stress (σ_{P1}) = $\frac{16}{\pi D^3} \left[M + \sqrt{M^2 + T^2} \right]$

$$= \frac{16}{\pi \times (0.1)^3} \left[1000 + \sqrt{1000^2 + 2000^2} \right]$$

$$= 1.64895 \times 10^7 \text{ kg/m}^2$$

$$= 1648.95 \text{ kg/cm}^2$$

Minor Principal stress (σ_{P2}) = $\frac{16}{\pi (0.1)^3} \left[1000 - \sqrt{1000^2 + 2000^2} \right]$

$$= -6.2984 \times 10^6 \text{ kg/cm}^2$$

$$\sigma_{P_2} = -629.84 \text{ kg/cm}^2$$

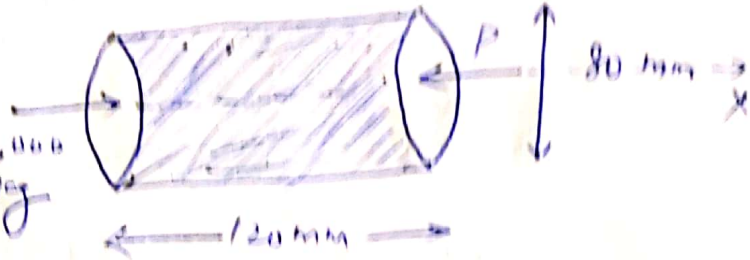
$$\tau_{\max} = \frac{\sigma_{P_1} - \sigma_{P_2}}{2} \quad \text{or} \quad \frac{16}{\pi D^3} \sqrt{M^2 + T^2}$$

$$\tau_{\max} = 1139.4 \text{ kg/cm}^2$$

$$Q-5(a) (0.1176 \text{ cm}^3)$$

Axial Compressive load (P) = 50,000 kg

$$P = 50,000 \text{ kg}$$



$$\text{Bulk Modulus (K)} = 1.7 \times 10^6 \text{ kg/cm}^2$$

$$\text{Poisson's ratio (}\nu\text{)} = 0.30$$

Let ϵ_v be the volumetric strain

$$\epsilon_v = \frac{\Delta V}{V} = \epsilon_x + \epsilon_y + \epsilon_z$$

$$\epsilon_x = \frac{\sigma_x}{E} - \nu \frac{\sigma_y}{E} - \nu \frac{\sigma_z}{E}$$

$$\epsilon_y = \frac{\sigma_y}{E} - \nu \frac{\sigma_z}{E} - \nu \frac{\sigma_x}{E}$$

$$\epsilon_z = \frac{\sigma_z}{E} - \nu \frac{\sigma_x}{E} - \nu \frac{\sigma_y}{E}$$

In this case, $\sigma_y = 0$, $\sigma_z = 0$

$$\sigma_x = \frac{P}{A} = \frac{-50,000}{\frac{\pi}{4} \times 8^2} = -975.2 \text{ kg/cm}^2$$

[-ve bcz its Compressive]

We also know that

$$E = 3K(1-2\nu)$$

$$E = 3 \times 1.7 \times 10^6 (1 - 2 \times 0.3)$$

$$E = 2.04 \times 10^6 \text{ kg/cm}^2$$

$$E_x = \frac{\sigma_x}{E} - \frac{\nu \sigma_y}{E} - \frac{\nu \sigma_z}{E} = 0$$

$$E_x = \frac{-995.2}{2.04 \times 10^6} = -4.878 \times 10^{-4}$$

$$E_y = -\frac{\nu \sigma_x}{E} = -0.3 \times (-4.878 \times 10^{-4})$$

$$= 1.4635 \times 10^{-4}$$

$$E_z = -\frac{\nu \sigma_x}{E} = 1.4635 \times 10^{-4}$$

$$E_v = E_x + E_y + E_z$$

$$= -4.878 \times 10^{-4} + 2 \times (1.4635) \times 10^{-4}$$

$$= -1.951 \times 10^{-4}$$

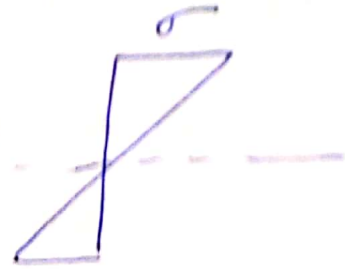
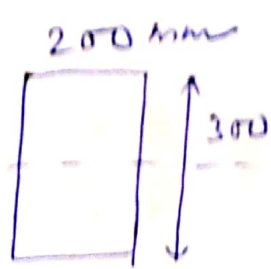
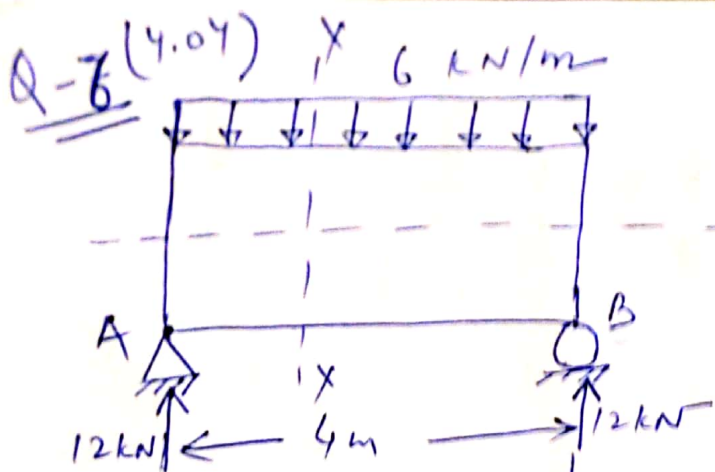
$$\frac{\Delta V}{V} = -1.951 \times 10^{-4}$$

$$\Delta V = (-1.951 \times 10^{-4}) \times \frac{\pi}{4} \times (8^2) \times 12$$

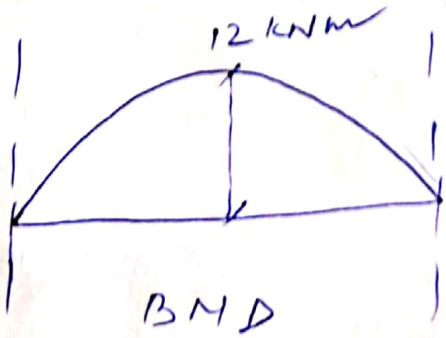
$$= -0.1176 \text{ cm}^3$$

Change in volume is -ve.

Volume reduction = 0.1176 cm³



Bending stress diagram



Bending moment Equation

$$(M_x) = 12 \cdot x - 6 \cdot x \cdot \frac{x}{2}$$

$$= 12x - \frac{6x^2}{2}$$

When $x=0$, $M_A = 0$

When $x=2m$, $M_{max} = 12 \times 2 - 3 \times 2^2$

$$= 24 - 12$$

$$= 12 \text{ kN-m}$$

Acc to Bending Equation

$$\frac{M}{I} = \frac{\sigma}{y} = \frac{E}{R}$$

(4.04)

$$\sigma = \frac{M \cdot y}{I} = \frac{M}{Z} = \frac{12 \times 10^6 \times 6}{200 \times 300^2}$$

$$= 4 \text{ N/mm}^2$$

$$\tau = 0.2 \text{ N/mm}^2$$

(Max principal stress)

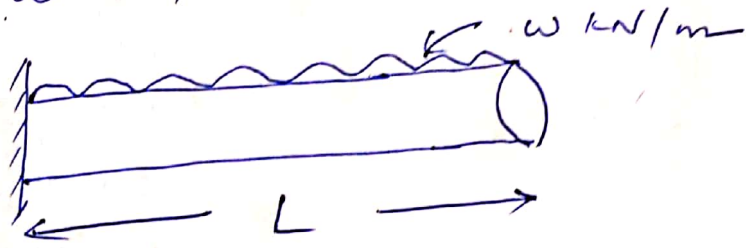
$$\sigma_{P_1} = \frac{\sigma}{2} + \sqrt{\left(\frac{\sigma}{2}\right)^2 + \tau^2}$$

$$= \frac{4}{2} + \sqrt{\left(\frac{4}{2}\right)^2 + 0.4^2} = 2 + \sqrt{4 + 0.4^2}$$

$$= 4.04 \text{ N/mm}^2$$

Q-7. (0.031, 32)

Let the intensity of loading & length of span be w kN/m and L respectively.



Max^m Curvature = $1.018592 \times 10^{-6} / \text{mm}$

Max^m Shear force = L kN

Atc to Bending Eqⁿ

$$\frac{M}{I} = \frac{\sigma}{y} = \frac{E}{R}$$

$$\left(\frac{1}{R} \right) = \frac{M}{EI}$$

Curvature

Max^m Curvature = $\frac{M_{\text{max}}}{EI}$

$$1.018592 \times 10^{-6} / \text{mm} = \frac{M_{\text{max}}}{2 \times 10^5 \times \frac{\pi}{64} (2w)^4}$$

$$M_{\text{max}} = 16 \times 10^6 \text{ N-mm}$$

$$M_{\text{max}} = 16 \text{ kNm}$$

Max^m B.M in cantilever shown above occurs at the fixed support of magnitude $wL^2/2$

$$\frac{wL^2}{2} = 16$$

$$wL^2 = 32 \quad \text{--- (1)}$$

Maxim shear force occurs at the fixed end (1)

End of magnitude = wL

$$wL = 1 \quad \text{--- (2)}$$

Dividing Eqn (1) by Eqn (2)

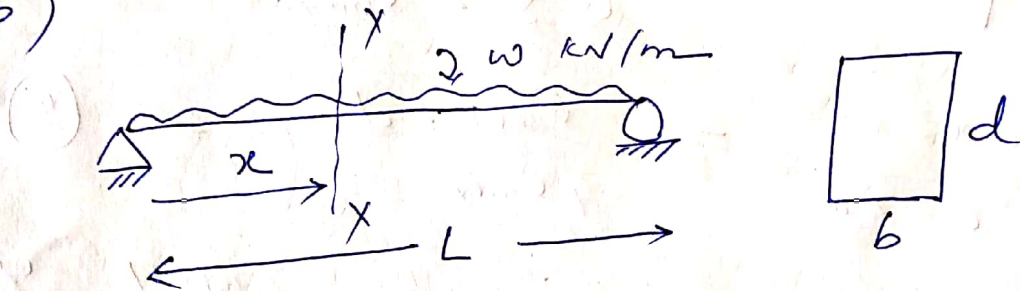
$$\frac{wL^2}{wL} = \frac{32}{1}$$

$$L = 32 \text{ m}$$

$$w = \frac{1}{32} \text{ kN/m} = 0.031 \frac{\text{kN}}{\text{m}}$$

$$w = \frac{1000}{32} = 31.25 \text{ N/m}$$

Q-8 (b)



Let w be the load intensity

$$\text{Given } \tau_{\text{max}} = 1.5 \frac{V}{bd} = \frac{3V}{2bd} \quad \text{--- (1)}$$

$$\text{Maxim bending stress } (\sigma_{\text{max}}) = \frac{M \times d/2}{\frac{bd^3}{12}} = \frac{6M}{bd^2} \quad \text{--- (2)}$$

Let M_x be the Bending Moment at $x-x$ & V be the shear force at $x-x$

$$M = \frac{wL}{2} \cdot x - \frac{wx^2}{2}$$

$$V = \frac{wL}{2} - wx$$

$$\frac{M}{V} = \frac{\left(\frac{WL}{2}x - \frac{Wx^2}{2}\right)}{\frac{WL}{2} - Wx} = \frac{Lx - x^2}{L - 2x} \quad (10)$$

$$\frac{M}{V} = \left(\frac{Lx - x^2}{L - 2x}\right) \quad \text{--- (3)}$$

From (1) $V = \frac{2 \tau_{max} \cdot bd}{3}$

From (2) $M = \frac{\sigma_{max} \cdot bd^2}{6}$

$$\frac{M}{V} = \frac{\sigma_{max} \cdot bd^2 \times 3}{6 \times 2 \tau_{max} \cdot bd}$$

$$\frac{Lx - x^2}{L - 2x} = \frac{\sigma_{max} \cdot d}{4 \tau_{max}} \quad \text{--- (4)}$$

Maximum value of Bending stress will occur at mid span.

$$\frac{WL^2}{8} = \sigma_{max} \cdot \frac{bd^2}{6}$$

$$\tau_{max} = \frac{3}{2} \frac{V}{bd}$$

$$\tau_{max} = \frac{3}{2} \times \frac{WL}{2 \cdot bd} = \frac{3}{4} \frac{WL}{bd}$$

$$\tau_{max} \cdot \frac{2}{3} (bd) = \frac{WL}{2}$$

$$\frac{Lx - x^2}{L - 2x} = \frac{WL^2/8}{\frac{WL}{2}} = \frac{L}{4}$$

$$L^2 - 2Lx = 4Lx - 4x^2$$

$$4x^2 - 6Lx + L^2 = 0$$

$$x = \frac{6L \pm \sqrt{36L^2 - 4 \times 4 \times L^2}}{8}$$

$$x = \frac{6L \pm \sqrt{20L^2}}{8}$$

$$x = \left(6 \pm 2\sqrt{5}\right) \frac{L}{8}$$

$$x = 1.309L, \quad 0.191L$$

the critical length will be 0.191L from end A.

Q-9 (4, 3)

$$y = \frac{1}{EI} \left(-2x^3 + \frac{x^4}{6} + 36x \right)$$

$$\frac{dy}{dx} = \frac{1}{EI} \left(-6x^2 + \frac{4x^3}{6} + 36 \right)$$

$$\frac{d^2y}{dx^2} = \frac{1}{EI} \left(-12x + \frac{12x^2}{6} \right)$$

$$EI \frac{d^2y}{dx^2} = -12x + 2x^2$$

$$EI \frac{dy}{dx} = M$$

$$M = 2x^2 - 12x$$

$$\frac{dM}{dx} = 4x - 12 = 0$$

$$\frac{dVx}{dx} = 4 = 0$$

Finding the position where $V_x = 0$
 $4x - 12 = 0$ $x = 3m$

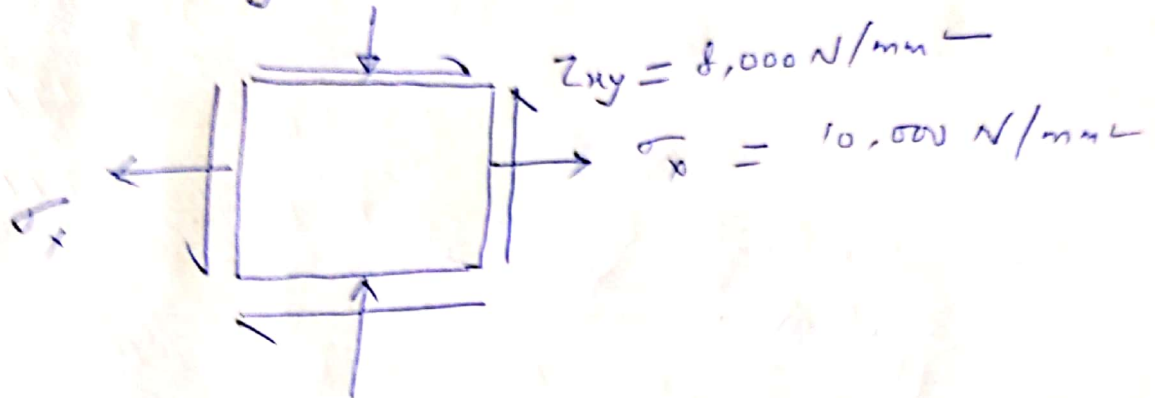
(12)

Q-10 (b)

$$\sigma_x = 10,000 \text{ N/mm}^2$$

$$\sigma_y = -6000 \text{ N/mm}^2$$

$$\tau_{xy} = 8000 \text{ N/mm}^2$$



$$\sigma_y = 6,000 \text{ N/mm}^2$$

Let σ_{p1}/σ_{p2} be the Major & Minor Principal Stress.

$$\begin{aligned} \sigma_{p1}/\sigma_{p2} &= \frac{\sigma_x + \sigma_y}{2} \pm \sqrt{\left(\frac{\sigma_y - \sigma_x}{2}\right)^2 + \tau_{xy}^2} \\ &= \frac{10,000 - 6,000}{2} \pm \sqrt{\left(\frac{-8000 - 10000}{2}\right)^2 + (8,000)^2} \\ &= 2000 \pm \sqrt{(8000)^2 + (8000)^2} \\ &= 2000 \pm 11313.71 \end{aligned}$$

$$\sigma_{p1} = 13313.71 \text{ N/mm}^2$$

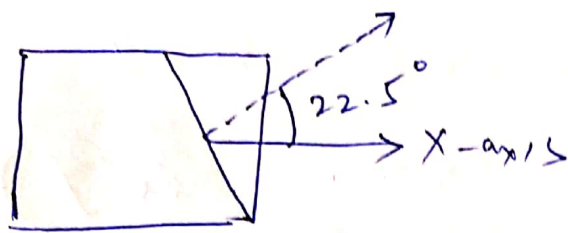
$$\sigma_{p2} = -9313.71 \text{ N/mm}^2$$

$$\tan 2\theta_p = \frac{2\tau_{xy}}{\sigma_x - \sigma_y} = \frac{2 \times 8000}{16000} = 1$$

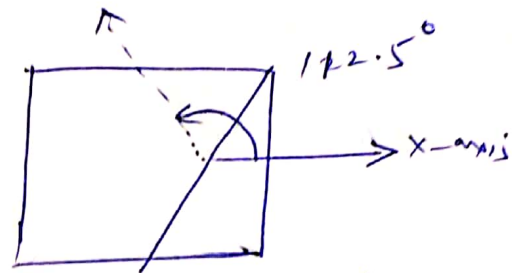
$$2\theta_p = 45^\circ$$

$$\theta_p = 22.5^\circ, \text{ or } 112.5^\circ$$

(13)



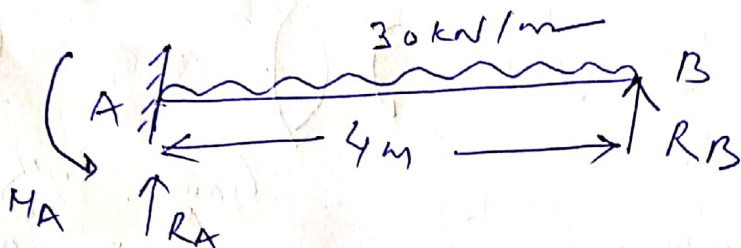
Major Principal Plane



Minor Principal Plane

Q-11 (0.115, 60)

$$EI = 2 \times 10^4 \text{ kN-m}^2$$

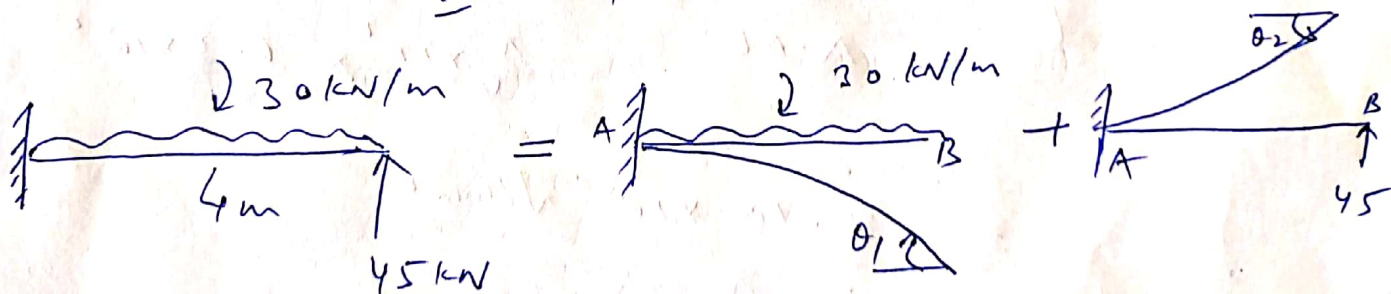


$$\delta_B = 0$$

$$\frac{wL^4}{8EI} - \frac{R_B L^3}{3EI} = 0$$

$$R_B = \frac{3wL}{8} = \frac{3 \times 30 \times 4}{8}$$

$$= 45 \text{ kN}$$



$$\theta_1 = \frac{wL^3}{6EI} = \frac{30 \times 4^3}{6 \times 2 \times 10^4} = \frac{2}{125} \text{ radian (Clockwise)}$$

$$\theta_2 = \frac{R_B L^2}{2EI} = \frac{45 \times 4^2}{2 \times 2 \times 10^4} = \frac{9}{500} \text{ radian (Anti-clockwise)}$$

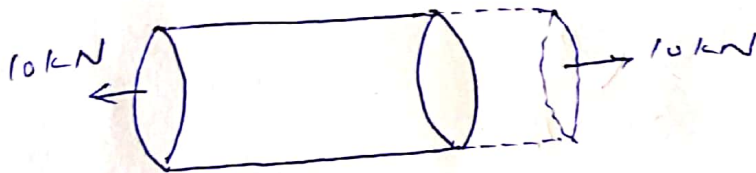
$$\theta_B = \theta_2 - \theta_1 = \frac{1}{500} = 0.002 \text{ radians} \quad (14)$$

(anticlockwise)

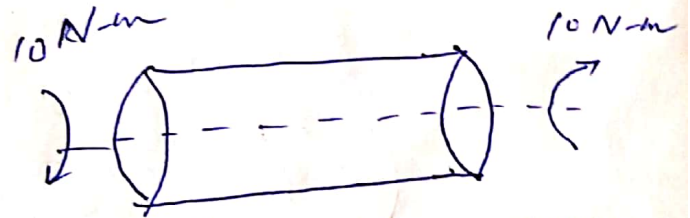
$$\begin{aligned} \text{Moment at A} &= 30 \times 4 \times \frac{4}{2} - 45 \times 4 \\ &= 60 \text{ kNm (Hogging)} \end{aligned}$$

$$\theta_B = 0.002 \times \frac{180}{\pi} = 0.115^\circ$$

Q-12 (2.10×10^5 , 1.373×10^5 , 8.44×10^4 , 0.245)



$$\begin{aligned} \text{Gauge length} &= 150 \text{ mm} \\ \Delta L &= 6.31 \times 10^{-2} \text{ mm} \\ \text{dia} &= 12 \text{ mm} \end{aligned}$$



$$\begin{aligned} \text{Gauge length} &= 150 \text{ mm} \\ \theta &= 0.5^\circ = \left(\frac{0.5\pi}{180} \right) \text{ radian} \\ &= 8.73 \times 10^{-3} \end{aligned}$$

$$\begin{aligned} \text{dia} &= 12 \text{ mm} \\ I_p &= \frac{\pi}{32} \times 12^4 = 2.0357 \times 10^3 \text{ mm}^4 \end{aligned}$$

$$\Delta = \frac{PL}{AE}$$

$$E = \frac{PL}{A \cdot \Delta L} = \frac{(10 \times 10^3) \times 150}{\frac{\pi}{4} \times 12^2 \times 6.3 \times 10^{-2}}$$

$$= 2.10 \times 10^5 \text{ N/mm}^2$$

Again from Torsion Data

$$\theta = \frac{TL}{N I_p} = \frac{(10 \times 10^3) \times 150}{(8.73 \times 10^{-3}) \times (2.03575 \times 10^3)}$$

$$N = 8.44 \times 10^4 \text{ mm}^2$$

$$E = 2N(1-\nu)$$

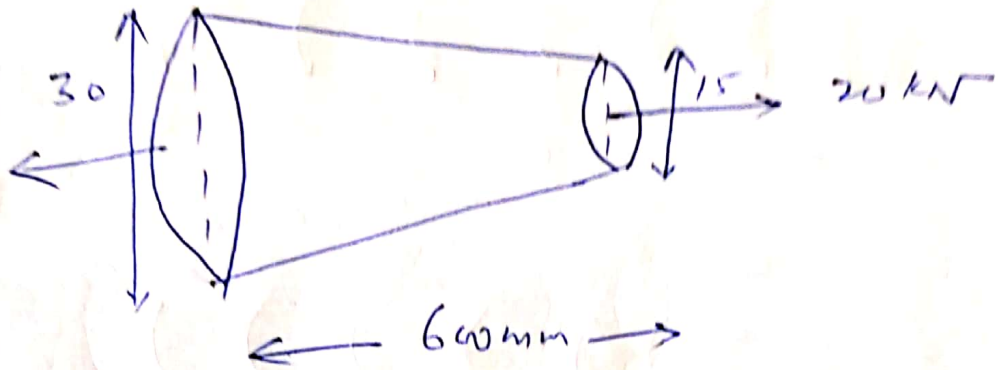
$$\nu = \left(\frac{E}{2N} - 1 \right) = 0.245$$

Again

$$E = 3L(1-\nu)$$

$$K = \frac{E}{3(1-\nu)} = 1.373 \times 10^5 \frac{N}{mm^2}$$

Q-13 (0.1697)



$$\Delta = \frac{4PL}{\pi E D_1 D_2} = \frac{4 \times 20 \times 10^3 \times 600}{\pi \times 15 \times 30 \times 2 \times 10^5}$$

$$\Delta = 0.1697 \text{ mm}$$

Q-14 (Marks to all)

$$E_x = +800, \text{ Take } \theta = 60^\circ$$

$$E_y = -1000, \quad \phi_{xy} = -60$$

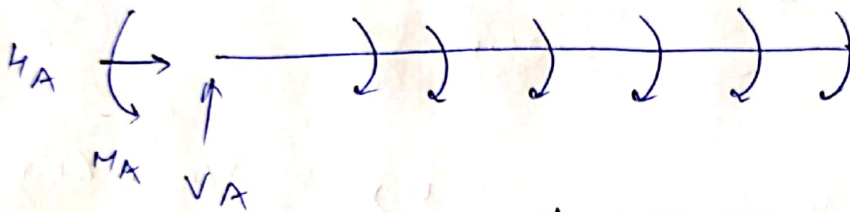
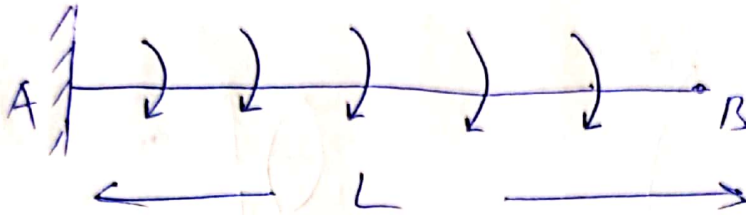
$$\begin{aligned} \epsilon_n &= E_x \cos^2 \theta + E_y \sin^2 \theta + \phi_{xy} \sin \theta \cos \theta \\ &= 800 \cos^2 60 - 1000 \sin^2 60 + (-60) \sin 60 \cos 60 \\ &= -809.81 \text{ (Contraction)} \end{aligned}$$

$$\frac{\phi_{x'y'}}{2} = (\epsilon_y - \epsilon_x) \sin^2 \theta + \frac{\phi_{xy}}{2} (\cos^2 \theta - \sin^2 \theta) \quad (16)$$

$$= -629.42$$

$$\phi_{x'y'} = -1258.84$$

Q-15 (0, ML)



$$\boxed{H_A = 0}$$

$$\boxed{V_A = 0}$$

$$-M_A + M \times L = 0$$

$$\boxed{M_A = ML}$$

